

THE HYDRODYNAMIC SOURCE TERM: BOUNDARY LAYER CUTTING AND FLOW CONDITIONING

S.G. DURBIN, S.I. ABDEL-KHALIK, and M. YODA

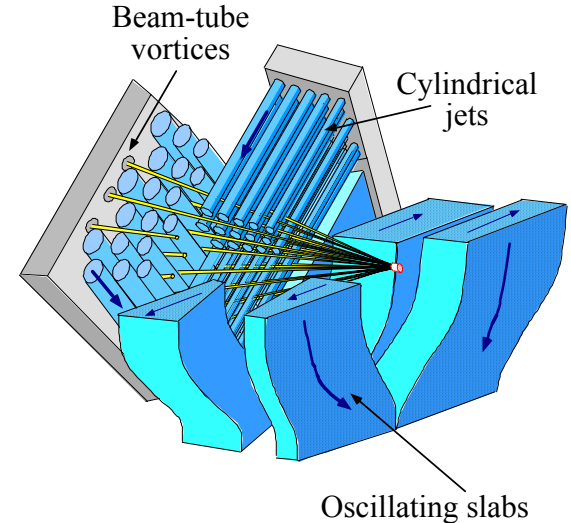
**G. W. Woodruff School of
Mechanical Engineering
Atlanta, GA 30332-0405 USA**



**Georgia Institute
of Technology**

RPD-2002 Chamber

- Use cylindrical jets and liquid sheets to shield IFE chamber first walls from neutrons, X-rays and charged particles
 - Oscillating slabs create protective pocket to shield chamber side walls
 - Lattice of stationary jets shield front/back walls while allowing beam propagation and target injection



Picture courtesy P.F. Peterson, UCB



RPD-2002 Jet Layout

Row	R (cm)	D _j (cm)	Row	R (cm)	D _j (cm)
0	90	4.61	6	155.0	8.95
1	98.07	5.15	7	170.6	10.0
2	107.1	5.75	8	188.1	11.2
3	117.1	6.42	9	207.7	12.5
4	128.4	7.18	10	229.5	13.9
5	140.9	8.01	11	253.9	15.6

- Jet speeds >12 m/s
- 252 jets total
- 12 rows/quad

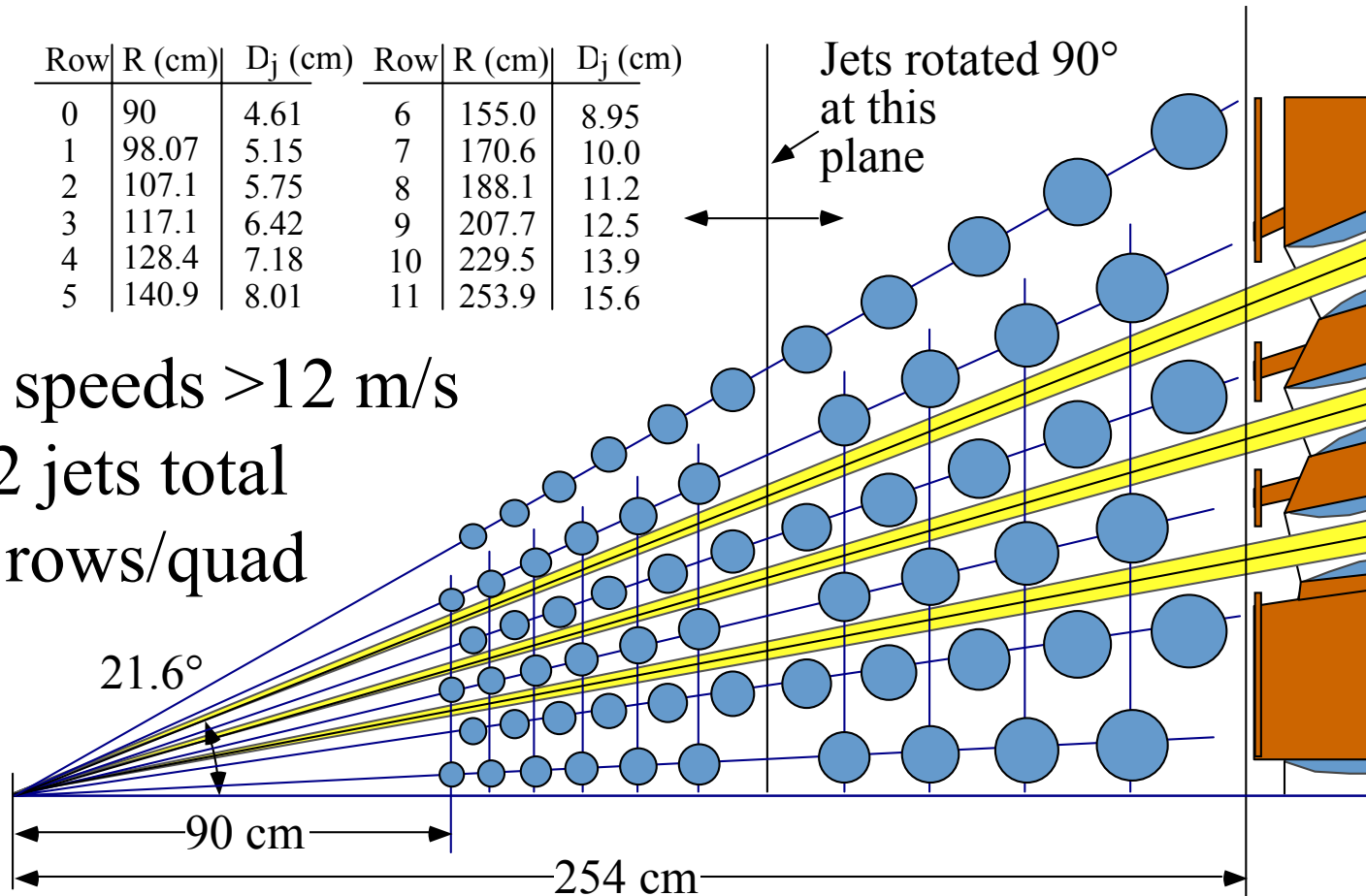
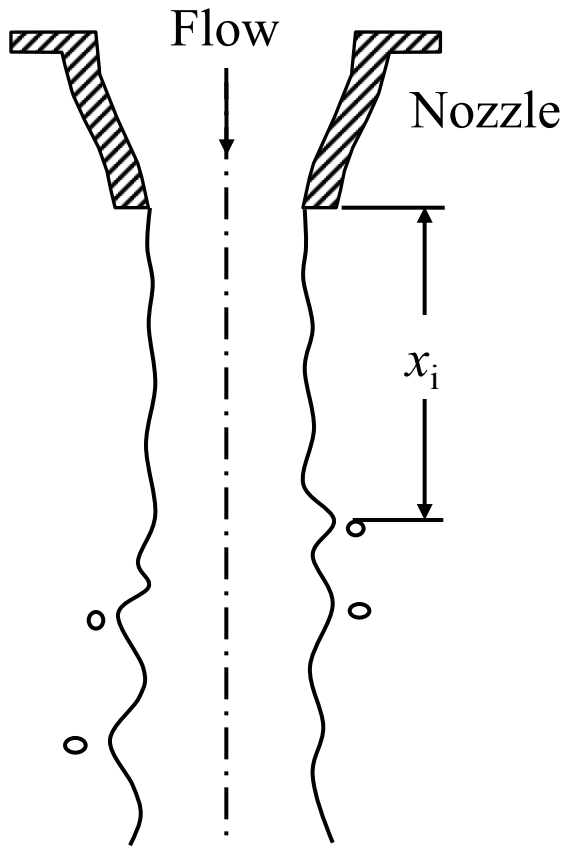


Diagram courtesy P.F. Peterson, UCB



Turbulent Breakup



- **Turbulent primary breakup**
 - Formation of droplets along free surface: “**hydrodynamic source term**”
 - Due to vorticity imparted at nozzle exit
- **Onset of breakup, x_i**
 - Location of first observable droplets
 - $x_i \downarrow$ as Weber number $We \uparrow$



RPD-2002 Correlation Results

[Sallam, Dai, & Faeth 2002]

- Total droplet mass ejection rate $\approx$ **1300 kg/s**
 - Assumes $G(x = 1 \text{ m})$ over entire surface area of each respective jet (Mean value of predictions)
 - $\sim 3\%$ of total jet mass flow rate
- Sauter mean dia. $\approx 5.7 \text{ mm}$ for all jets at $x = 1 \text{ m}$
 - SMD at $x_i \approx 0.82 - 1.0 \text{ mm}$ for $d = 4.61 - 15.6 \text{ cm}$, respectively



Comments

- **Evaporation rate due to target energy deposition assuming 6 Hz**

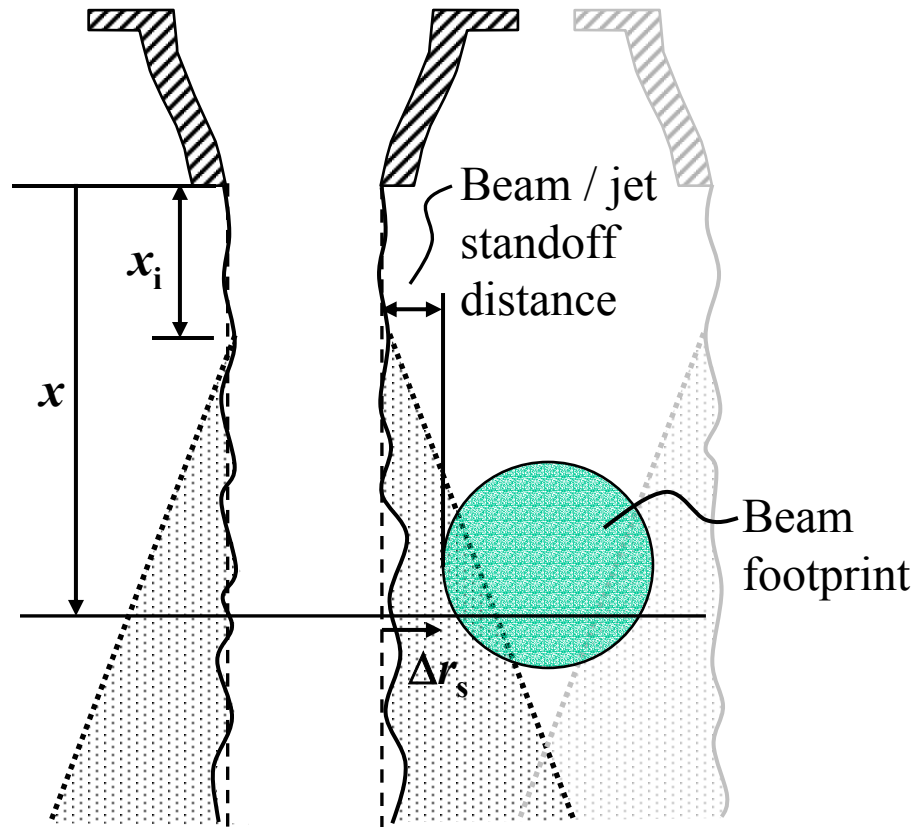
	Total (MJ)	X-rays and ions (MJ)	Mass rate (kg/s)
NRL - DD	154	45	37 - 51
HIF - ID	458	142	116 - 160

* Saturated Flibe: $\Delta h = h_{fg} = 5.3 \text{ MJ/kg}$
Subcooled Flibe: $\Delta h = 7.3 \text{ MJ/kg}$

- **Hydrodynamic source term ~1300 kg/s !**
 - Must be reduced (flow conditioning, nozzle, BL cutting, etc.)

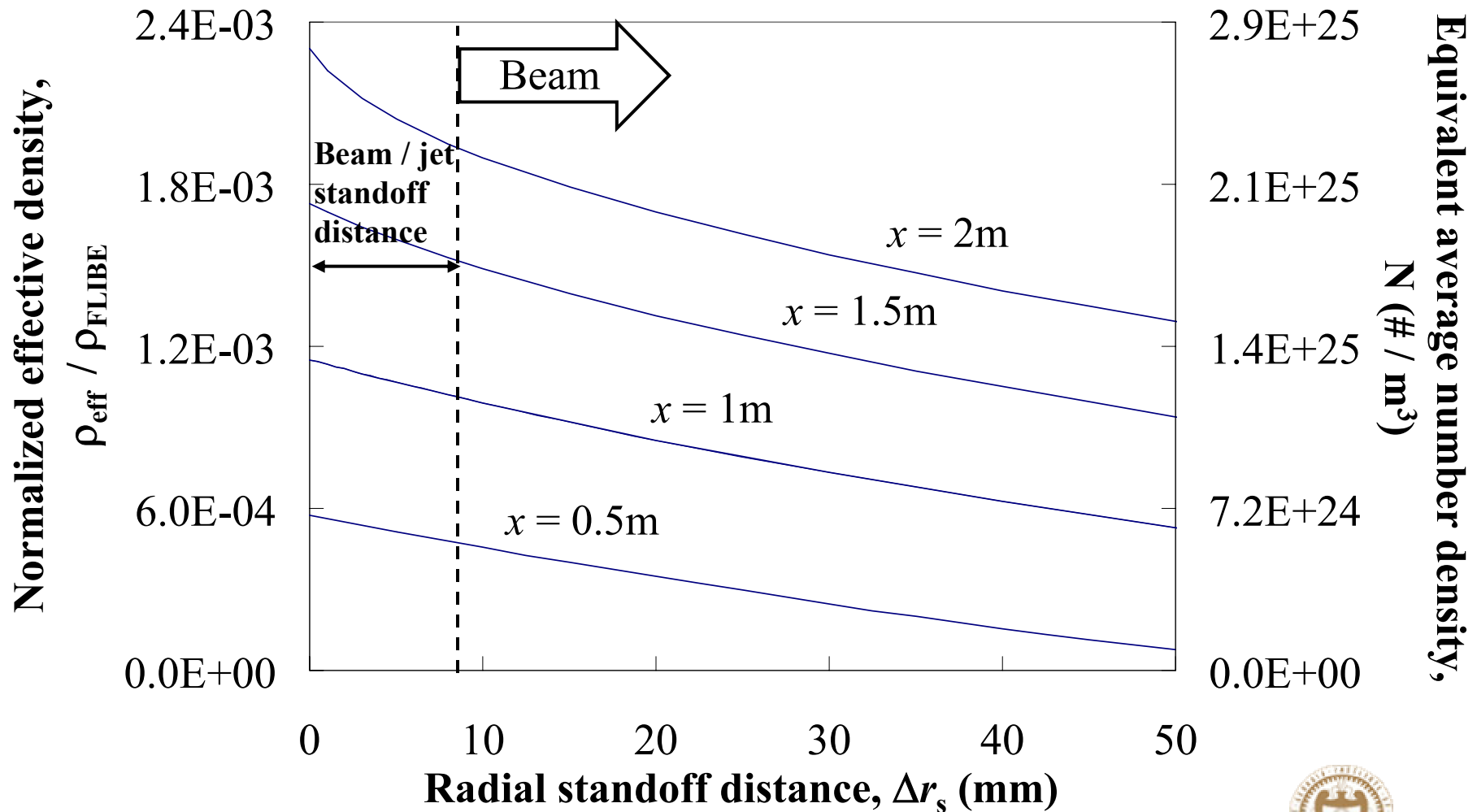


Implications for Beam Propagation

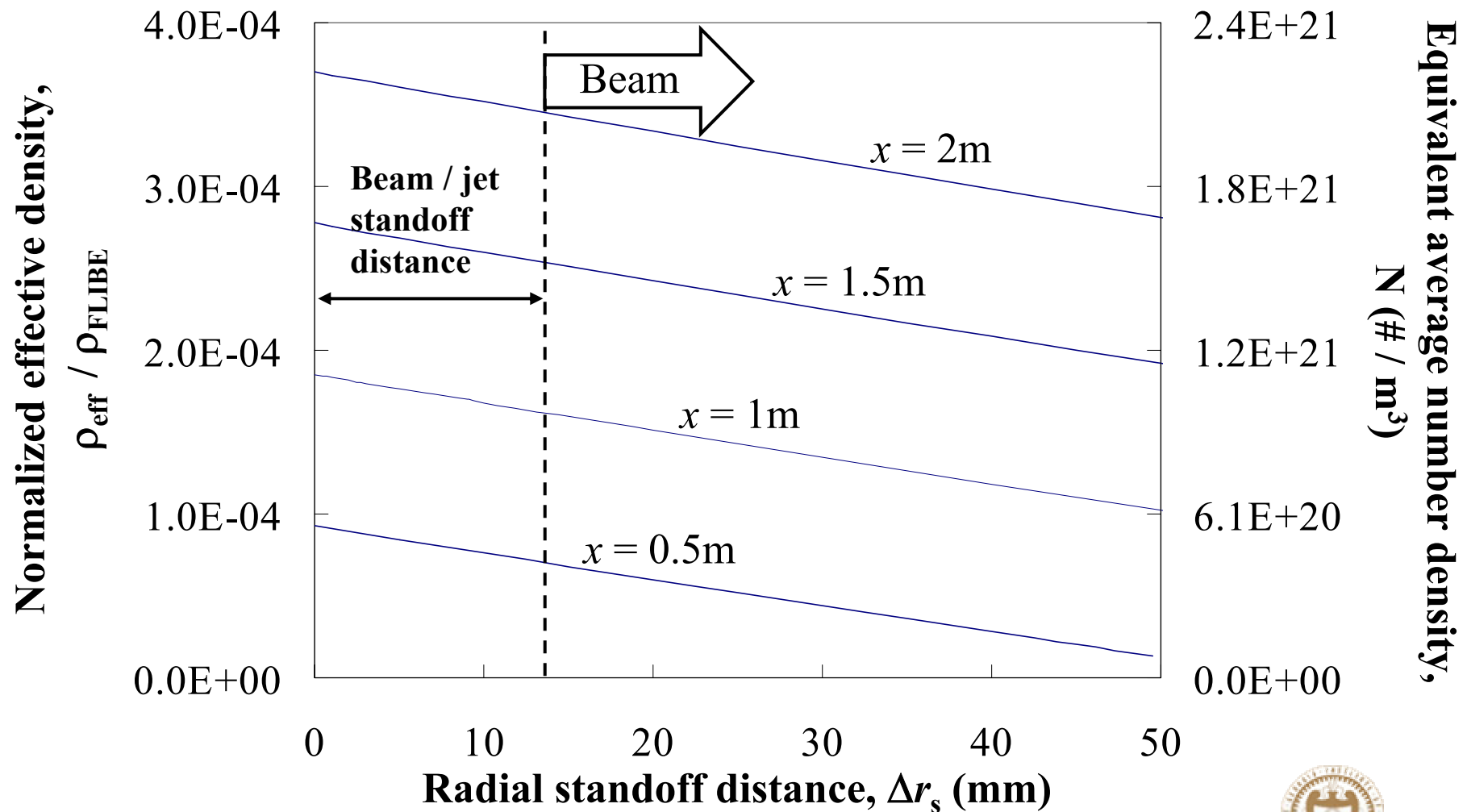


- **Droplets enter into beam footprint**
- **Radial standoff, Δr_s**
 - Measured from nominal jet surface
- **Equivalent number density dependent on x and Δr_s**
 - Ignores jet-jet interactions

Implications for Typical RPD-2002 Jet: $d = 4.61$ cm (Row 0)



Implications for Typical RPD-2002 Jet: $d = 15.6$ cm (Row 11)



Beam Propagation Implications

- **Model predictions imply protection concept is incompatible with beam propagation requirements**
- **However, model is based on :**
 - **Fully developed turbulent pipe flow at exit**
 - **No flow conditioning, nozzle or BL cutting**
- **Can nozzles / jets be designed to reduce these number densities to a level compatible with beam propagation requirements?**

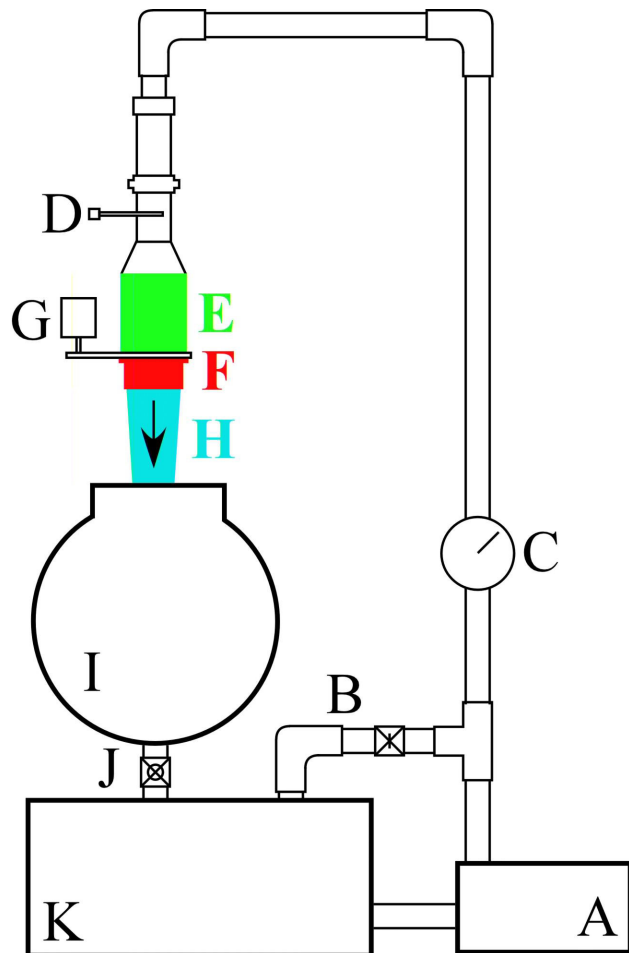


Objectives

- **Estimate rate of droplet formation due to hydrodynamics (turbulent breakup) for the thick liquid protection concept**
- **Evaluate effectiveness of boundary layer cutting**
 - **Can the hydrodynamic source term be reduced / eliminated?**
 - **Can BL cutting be used to reduce the hydrodynamic source term?**
 - **Can BL cutting be used in lieu of “traditional” flow conditioning?**



Flow Loop



- Pump-driven recirculating flow loop
- Test section height ~ 1 m
- Overall height ~ 5.5 m

A Pump

C Flow meter

E Flow conditioner

F Nozzle

H Sheet

J Butterfly valve

B Bypass line

D Pressure gage

G Oscillator (Not used)

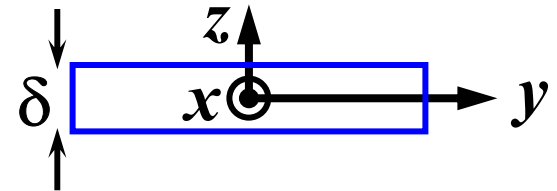
I 400 gal tank

K 700 gal tank

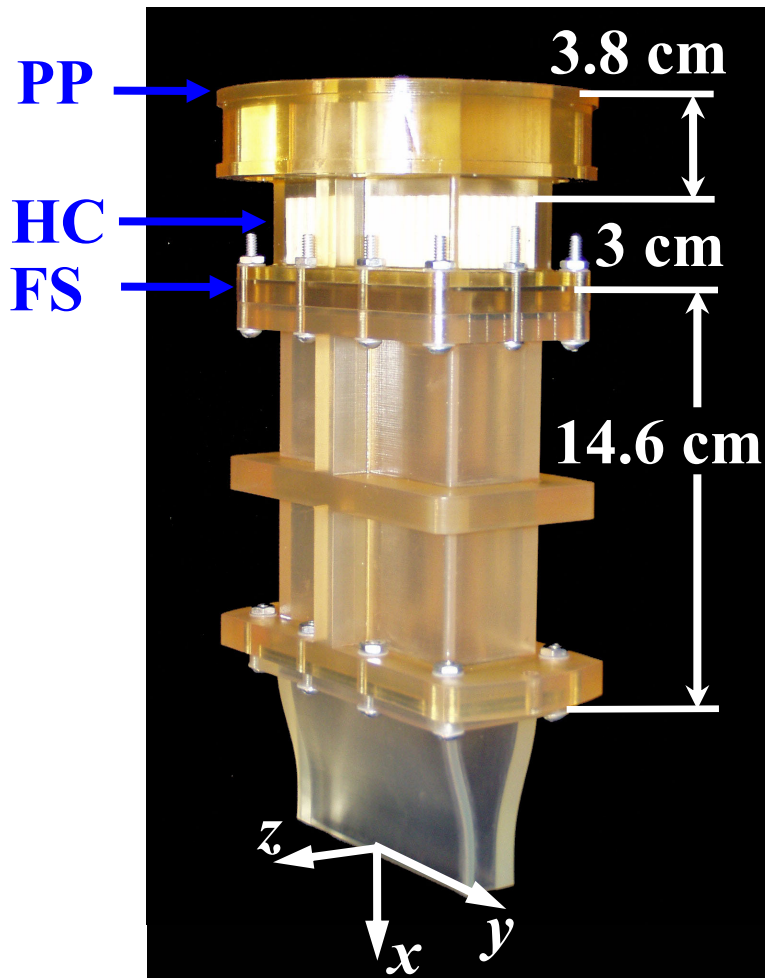


Experimental Parameters

- $\delta = 1 \text{ cm}$; aspect ratio $AR = 10$
- Near-field: $x / \delta \leq 25$
- Reynolds number $Re = 130,000$ [$Re = U_o \delta / \nu$; U_o average speed; ν liquid kinematic viscosity]
 - $Re = 138,000 - 467,000$ for RPD-2002
- Weber number $We = 19,000$ [$We = \rho_L U_o^2 \delta / \sigma$; ρ_L liquid density; σ surface tension]
 - $We = 68,000 - 230,000$ for RPD-2002
- Froude number $Fr = 1,400$ [$Fr = U_o^2 / (g\delta)$; g gravitational acceleration]
- Fluid density ratio $\rho_L / \rho_G = 850$ [ρ_G gas density]



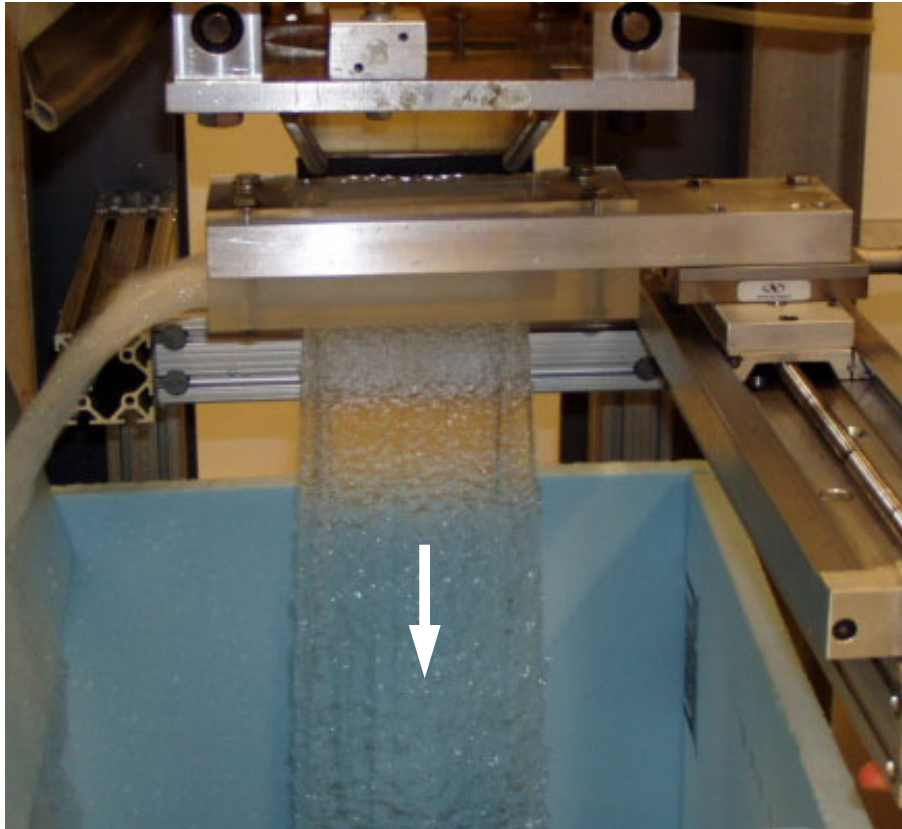
Flow Conditioning



- **Round inlet (12.7 cm ID) to rectangular cross-section 10 cm $\times$ 3 cm ($y \times z$)**
- **Perforated plate (PP)**
 - Open area ratio 50% with staggered 4.8 mm dia. holes
- **Honeycomb (HC)**
 - 3.2 mm dia. $\times$ 25.4 mm staggered circular cells
- **Fine mesh screen (FS)**
 - Open area ratio 37.1%
 - 0.33 mm dia. wires woven w/ open cell width of 0.51 mm (mesh size 30 $\times$ 30)
- **5th order contracting nozzle**
 - Contraction ratio = 3



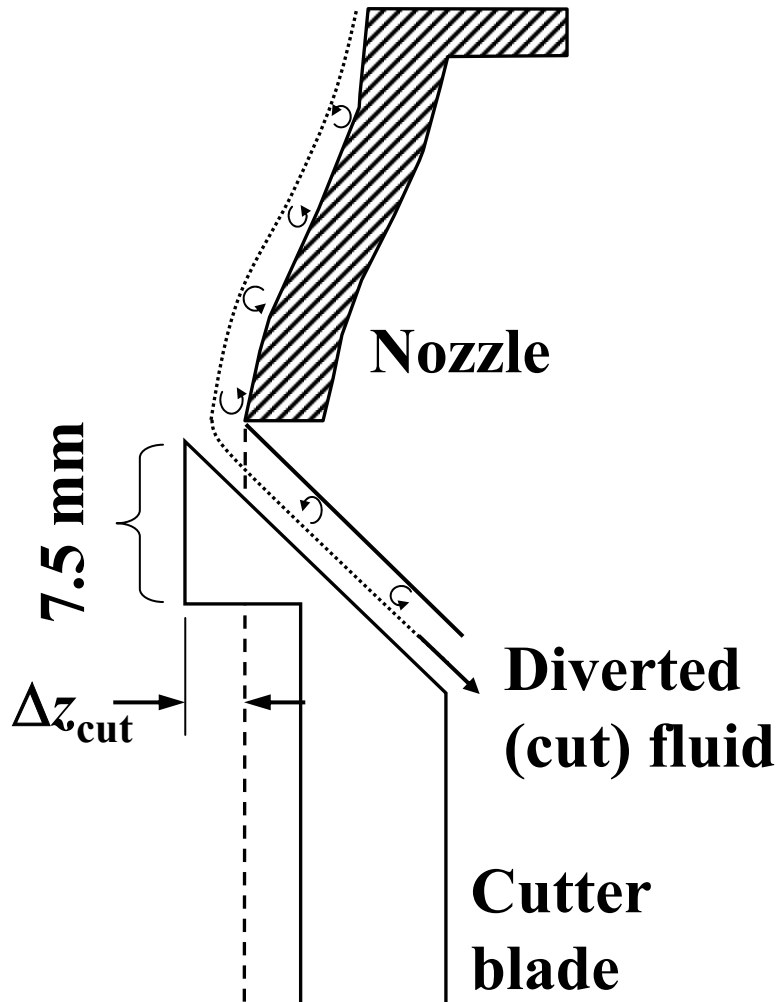
Boundary Layer Cutter



- “Cut” (remove BL fluid) on one side of liquid sheet
- Independently control:
 - Cut depth, Δz_{cut}
 - Downstream location of cut, x
- Removed liquid (~ 0.18 kg/s) diverted to side



Cutter Details

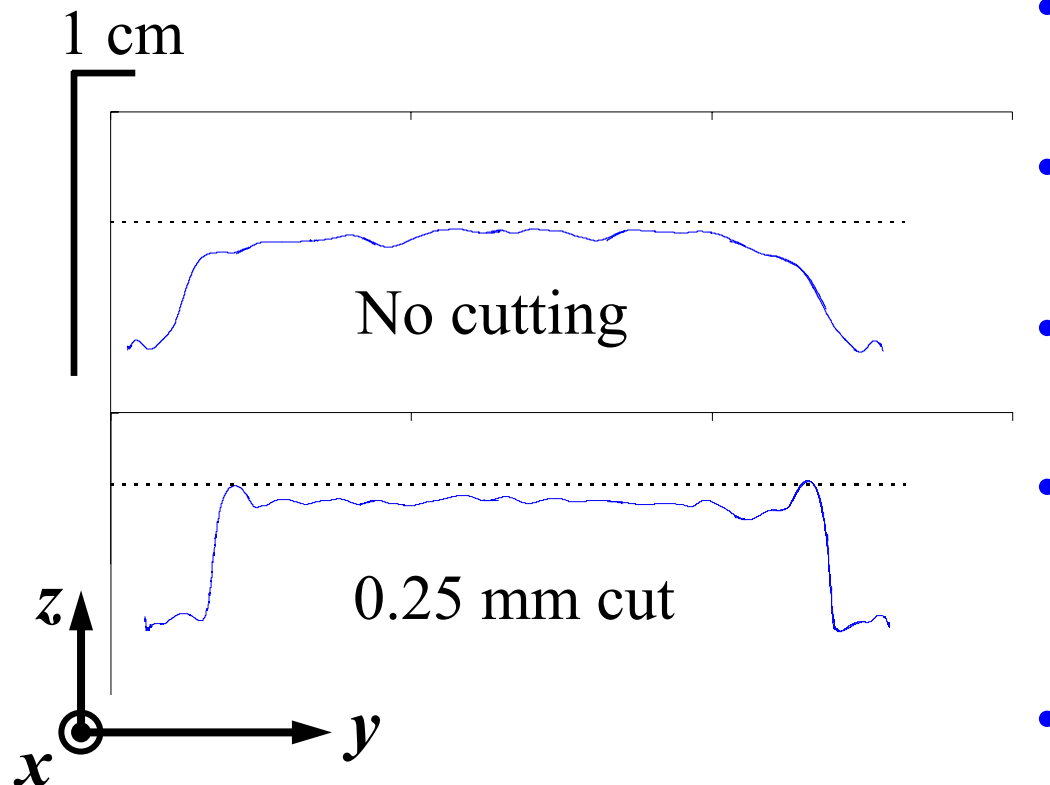


- **Aluminum blade inserted into flow**
 - Remove high vorticity / low momentum fluid near nozzle wall
 - Blade face tilted 0.4° from vertical
 - Blade width (y-extent) 12 cm
- **Relatively short reattachment length**
 - Nozzle contraction length 63 mm



Jet Profiles

$(x / \delta = 15)$



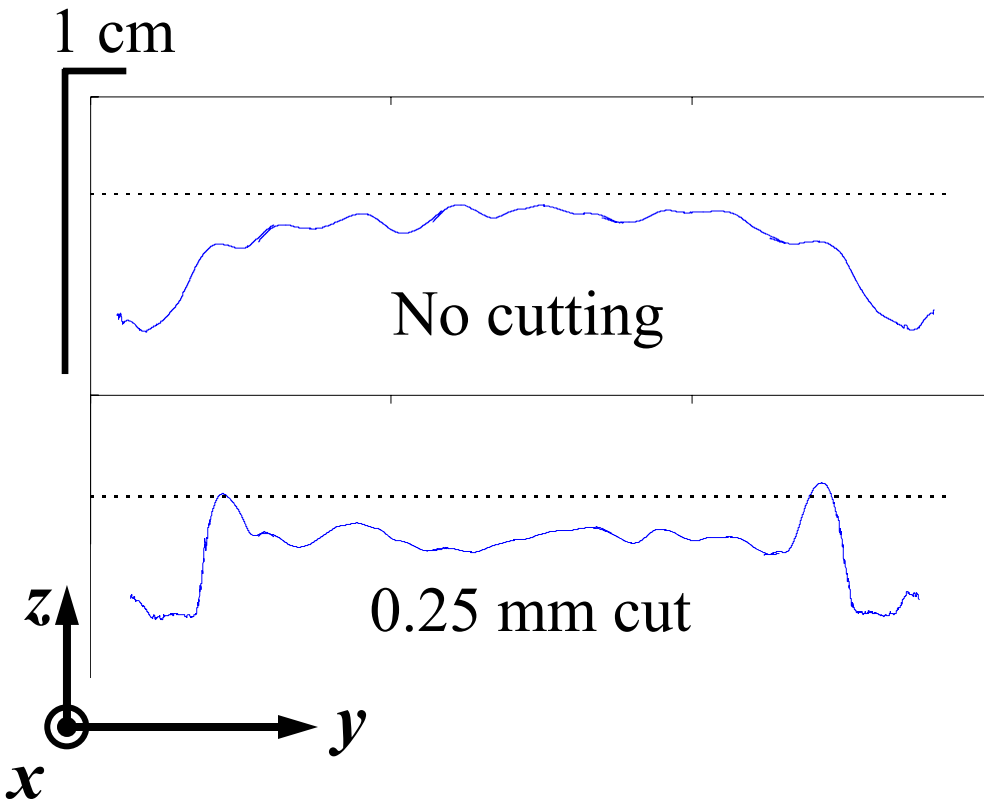
Note: Vertical axis at $5\times$ magnification

- Obtained from average of 130 images over 4.3 s
- Standard flow conditioning design
- Cut depth, $\Delta z_{\text{cut}} = 0.25$ mm
- Nominal free surface (jet nozzle) indicated by dashed lines
- BL cutting results in “dog-bone” structures near edges of jet



Jet Profiles

$(x / \delta = 25)$



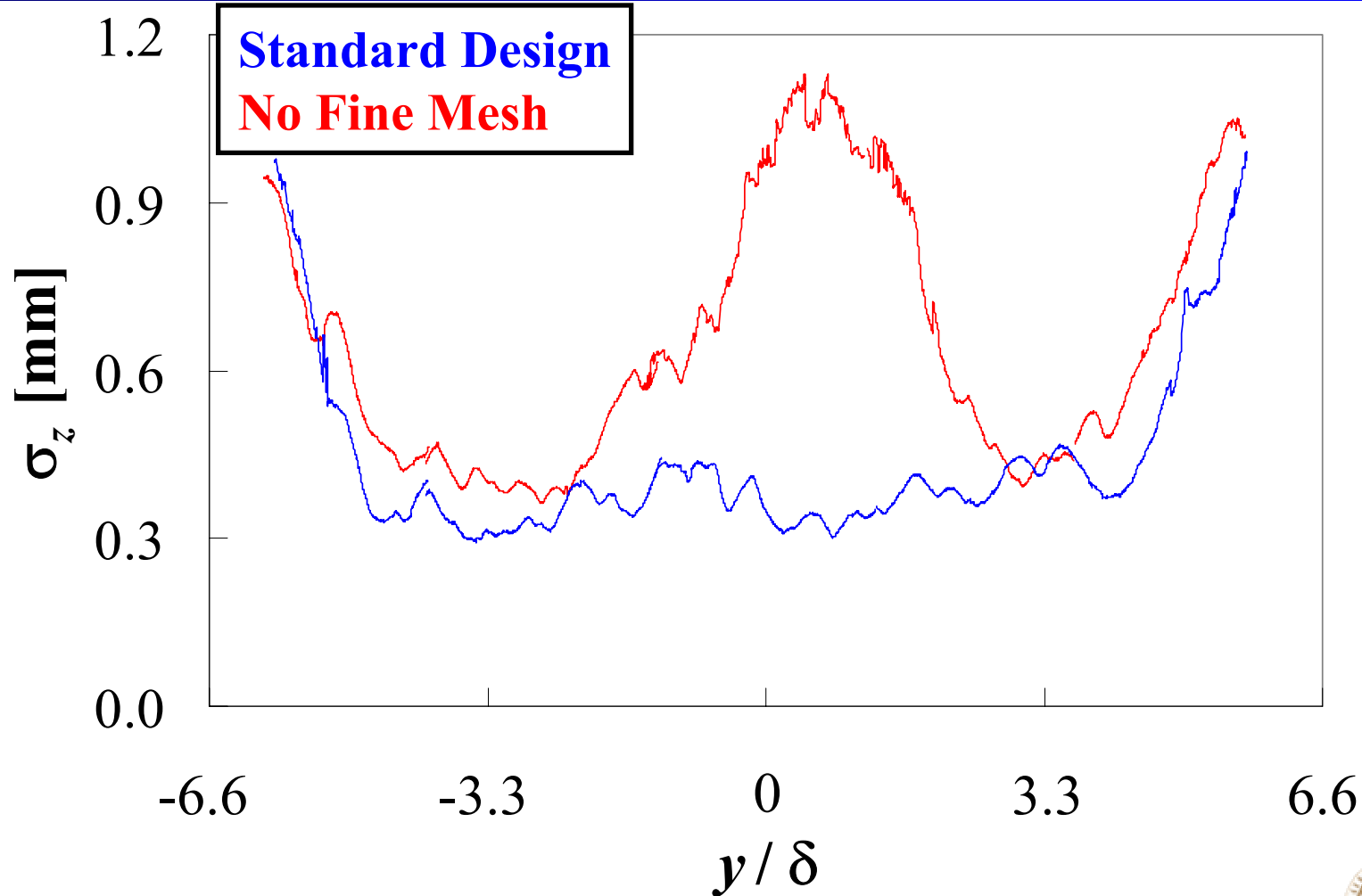
Note: Vertical axis at $5\times$ magnification

- **Uncut jet inside nominal free surface**
- **“Dog-bone” structures more pronounced**
 - **Sharp transition to edges of jet**
- **Jet width (y -extent) decreases with cutting**



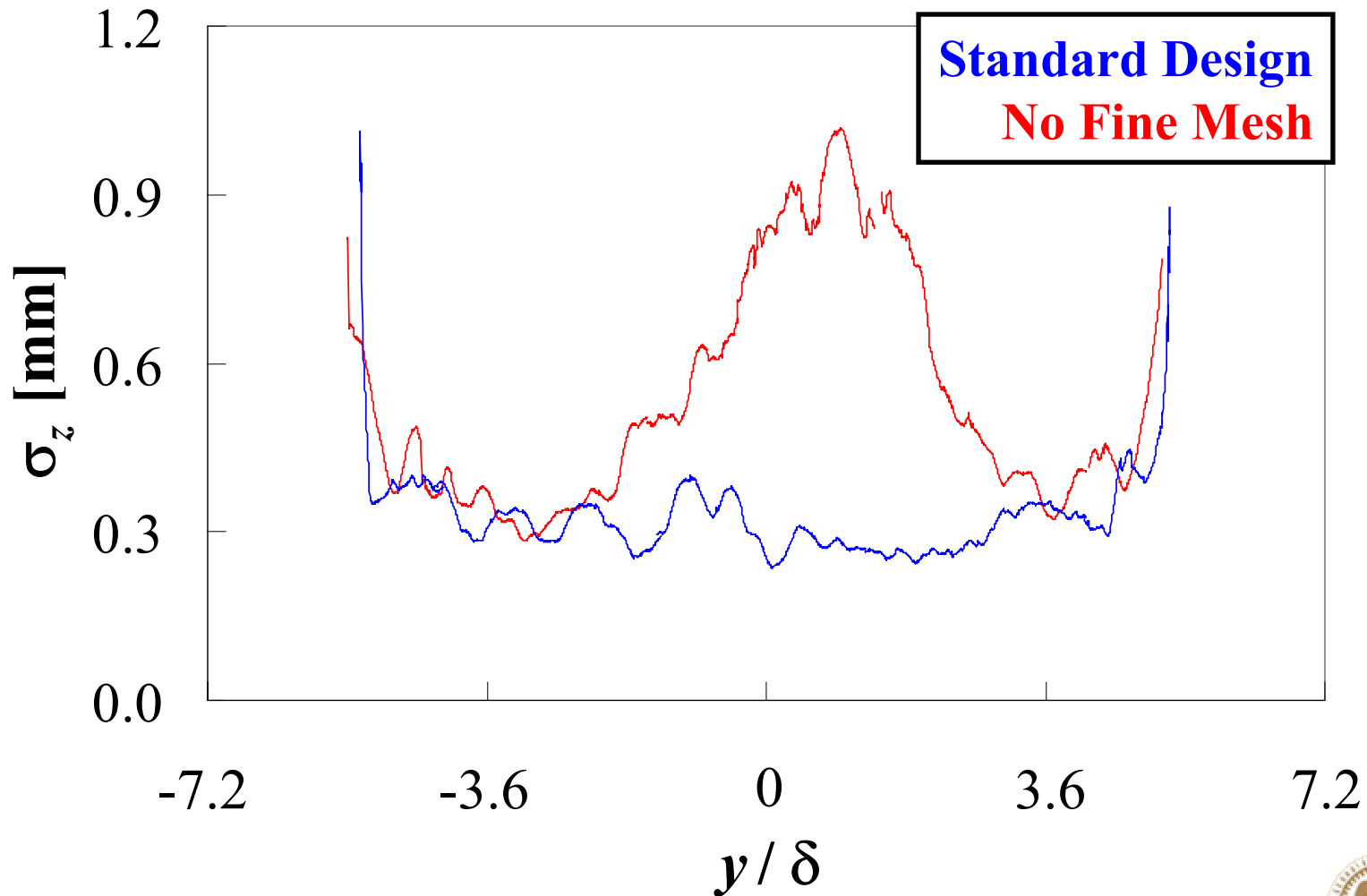
Flow Conditioning Effects

($x / \delta = 25$)

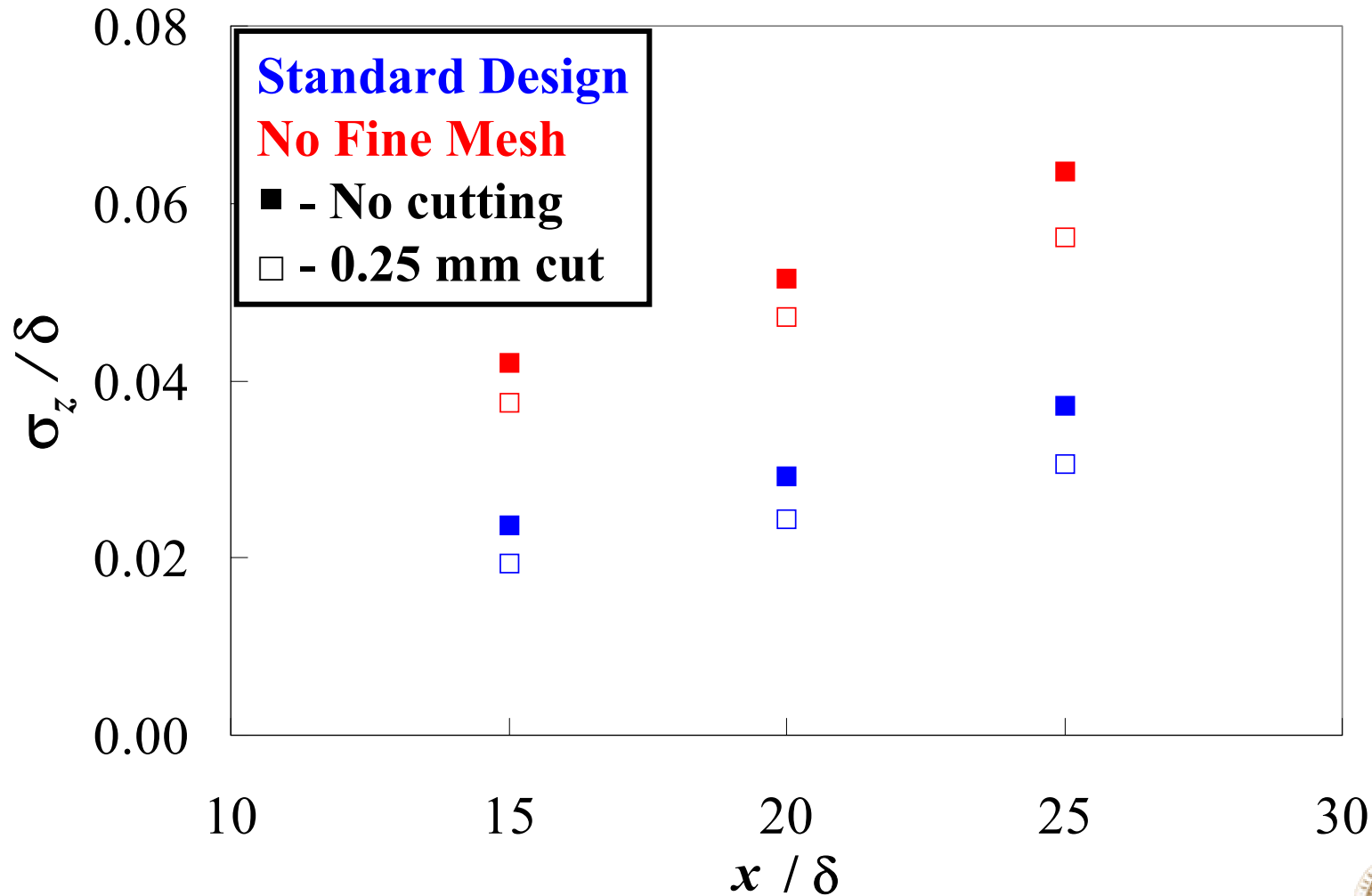


BL Cutting Effects

$(x / \delta = 25; \Delta z_{\text{cut}} = 0.25 \text{ mm})$



Average* Surface Ripple for $Re = 130k$



*Averaged over central 75% of flow, excludes “dog-bone” regions

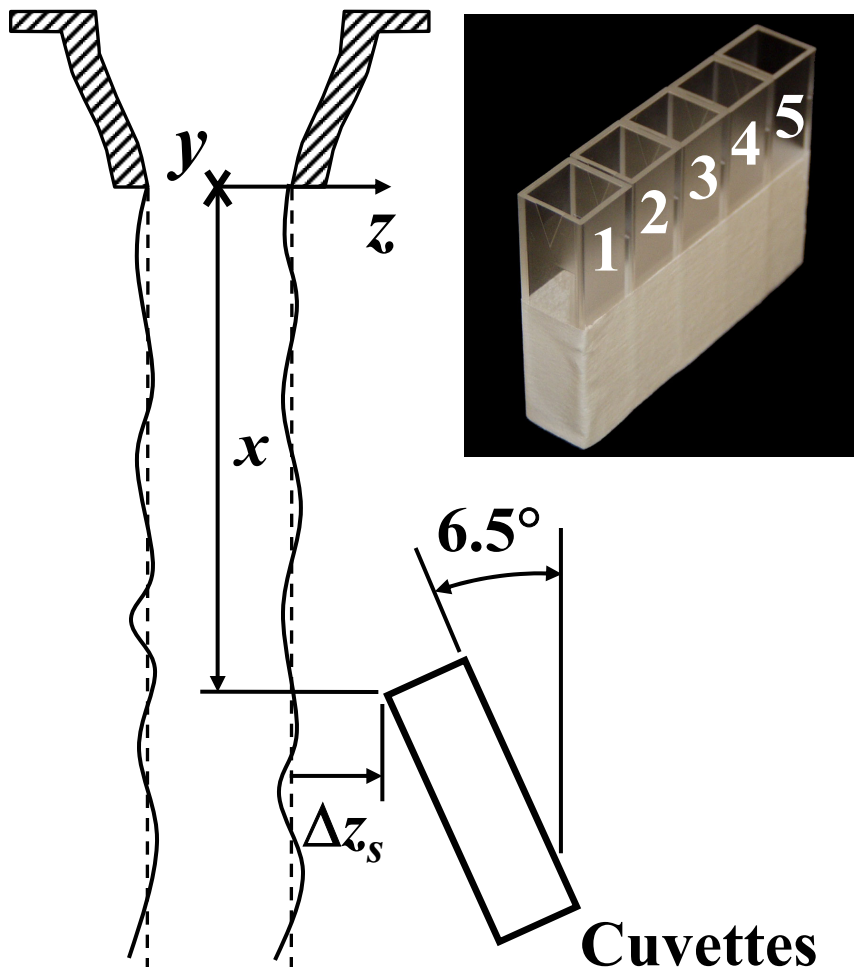


Summary: BL Cutting

- Reduces surface ripple by 17% for $\Delta z_{\text{cut}} = 0.25$ mm and standard flow conditioner
 - Preliminary data suggest 48% reduction in surface ripple for $\Delta z_{\text{cut}} = 0.38$ mm ($Re = 97,000$; $x / \delta = 16$)
- Re-emergence of “dog-bone” structures as seen at lower Re
- Jet width (y -extent) decreased by ~ 6 mm at $x / \delta = 25$



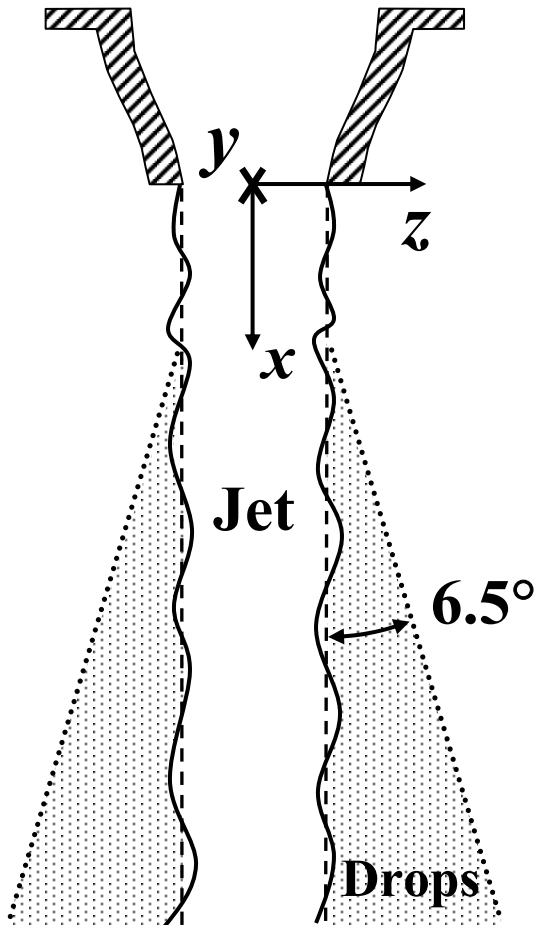
Mass Collection Procedure



- Cuvette opening = $1 \text{ cm} \times 1 \text{ cm}$ w/0.9 mm wall thickness
- Five adjacent cuvettes
 - Cuvette #3 centered at $y = 0$
- Located at x , Δz_s away from nominal jet position
 - $\Delta z_s \cong 2.5\text{--}15 \text{ mm}$
 - Experiments repeated to determine uncertainty in data
- Mass collected over 0.5–1 hr



Droplet Trajectories



- Droplets follow ballistic path based on $\tilde{u}$ and $\tilde{v}$
 - Absolute streamwise and radial velocities
$$\tilde{u} = 0.78 \cdot U_o, \quad \tilde{v} \leq 0.088 \cdot U_o$$
 - Neglects gravitational and aerodynamic effects
- Droplet “halo” forms starting at x_i
 - Droplets only inside halo



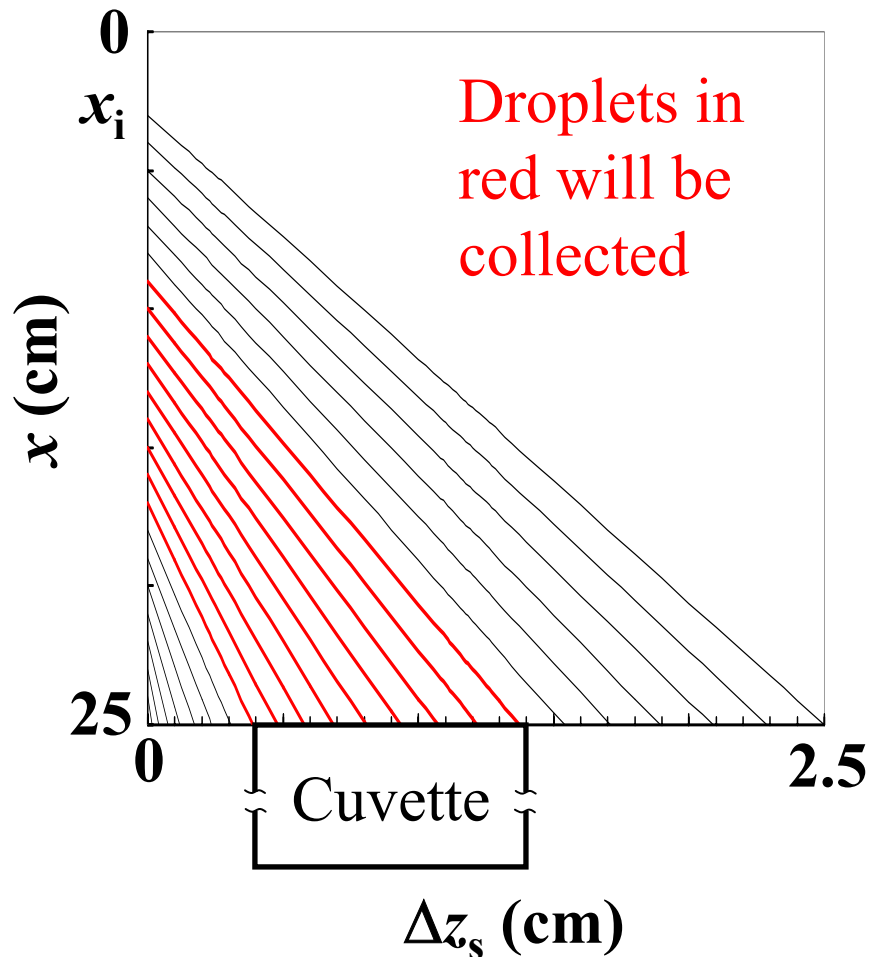
Droplet Visualizations



- Model predicts SMD = 0.6 – 1.3 mm
 - No flow conditioning
- Droplets collected on coated microscope slide
 - Standard flow conditioning
 - $Re = 130,000$; $x/\delta = 25$
 - Diameters $O(1-100 \mu\text{m})$
- Initial visualizations for no fine mesh suggest intermittent ejection of large (~ 1 mm dia.) drops



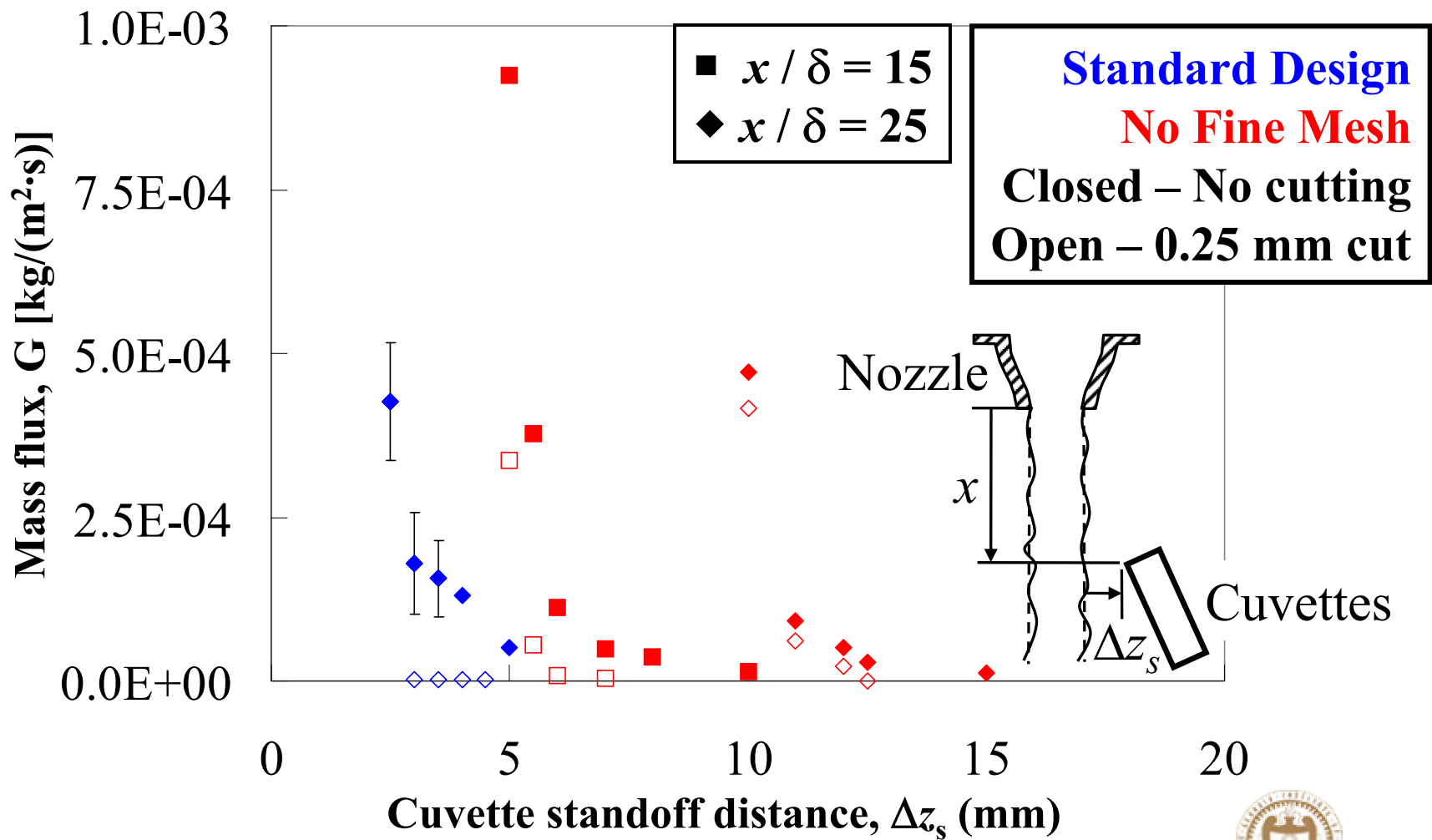
Collected Droplets



- Droplets with **red** trajectories collected by cuvettes
 - Cuvette shown at $x = 25$ cm, $\Delta z_s = 4$ mm, $\theta = 0^\circ$
- Note that $z = 0$ corresponds to nominal free surface of jet

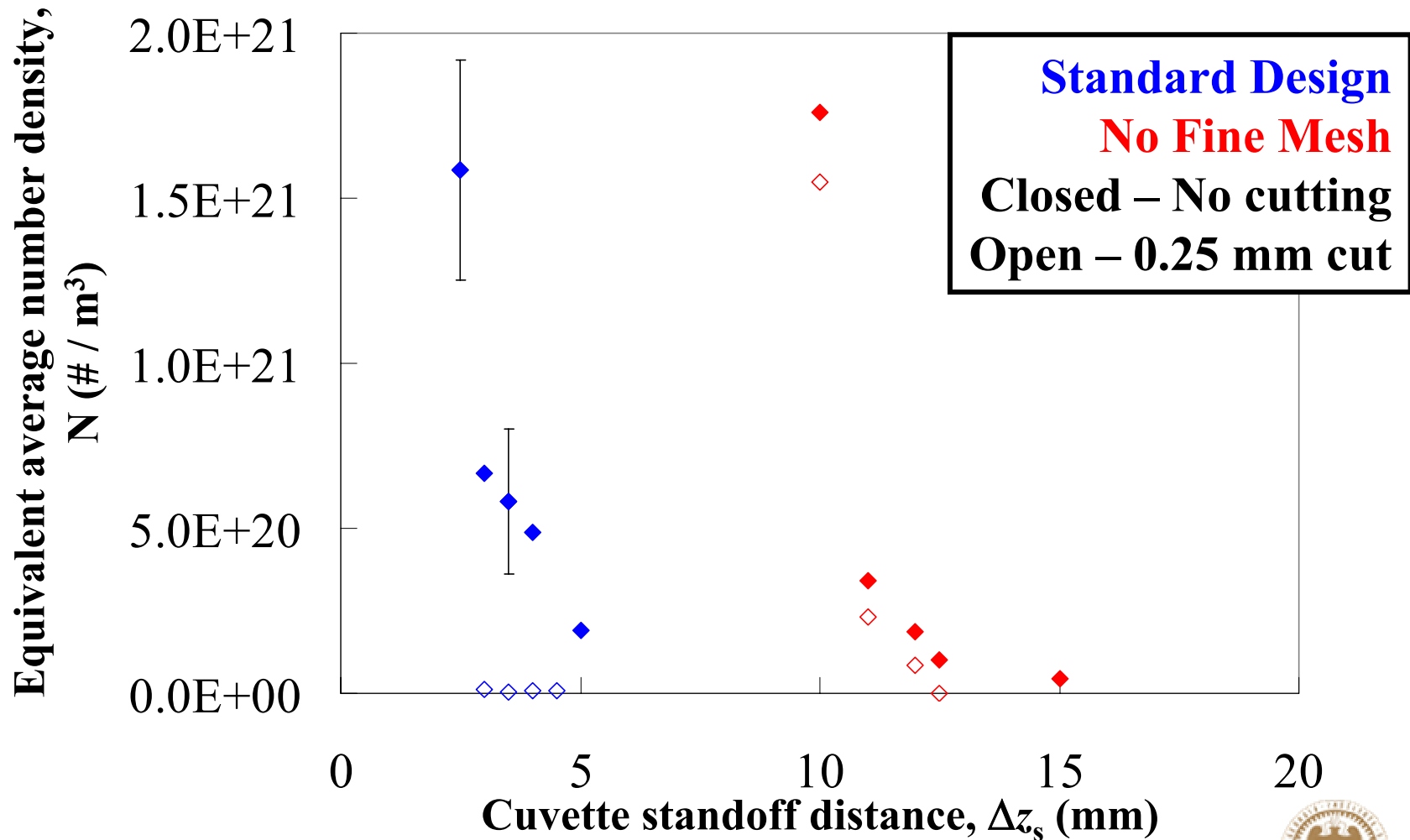


Experimental Mass Flux

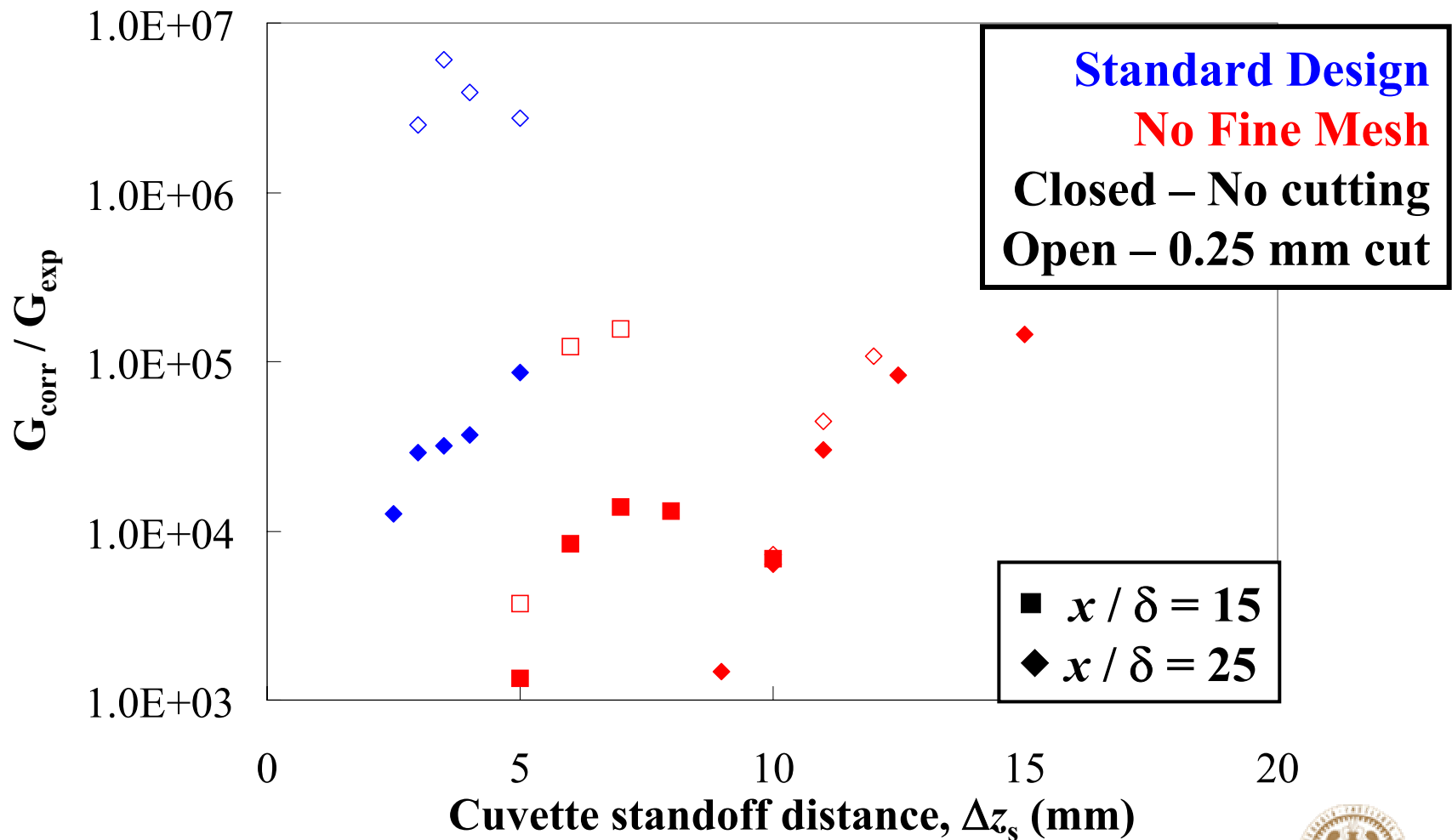


Experimental Number Density

$(x / \delta = 25)$



Model Comparison



Summary: Mass Collection

- Flow straightening and contracting nozzle significantly reduce ejected droplet mass (by **3–5 orders of magnitude**) compared w/model
- BL cutting has considerable impact on collected droplet mass
- BUT: proper flow conditioning more important
- Flow conditioning and BL cutting reduce collected droplet mass by orders of magnitude (compared with model predictions)



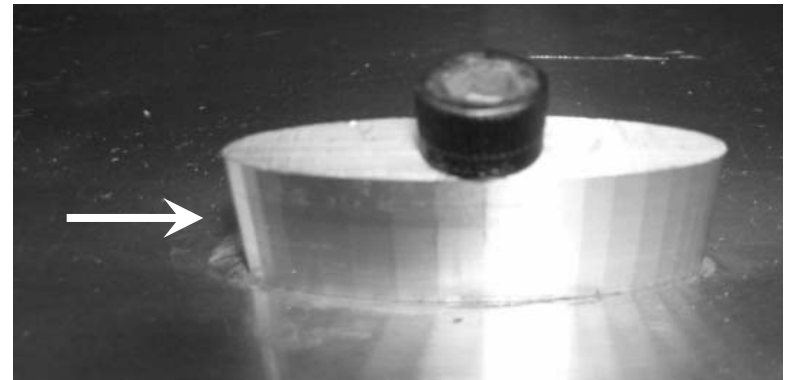
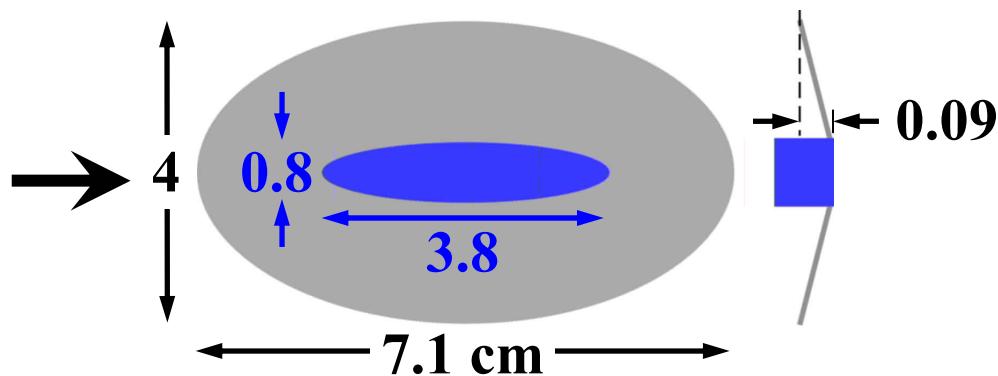
Conclusions

- Hydrodynamic source term sensitive to initial conditions
- Jet geometry, surface ripple and breakup affected by flow conditioning
- Flow conditioning / converging nozzle reduces droplet mass flux (and number density) by 3–5 orders of magnitude over model predictions
- BL cutting appears to eliminate droplet ejection for a “well-conditioned” jet
- Preventing blockage of fine mesh screens major issue

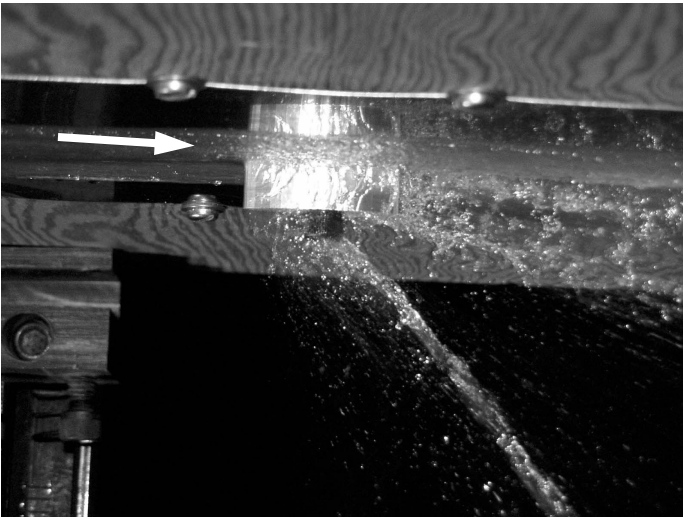


Hydrodynamically Shaped Penetration With Backwall Modification

- Scaled penetration/backwall geometry shown in APEX study to ensure fluid closure downstream of penetration
 - Simulations for perfectly wetting fluid (0° contact angle)
 - Leading edge $x = 16.4$ cm from nozzle exit
- Downward and upward-facing flat surfaces
 - No centrifugal acceleration



Initial Results



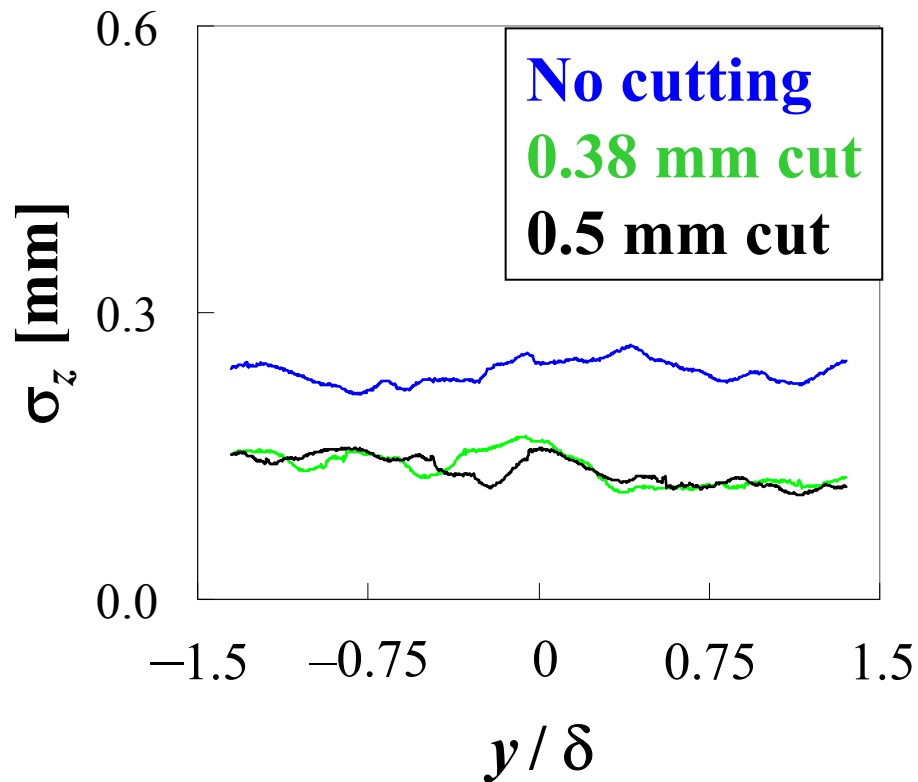
- Water on stainless steel (contact angle $\sim 50^\circ$)
- $Re = 10,000$; $We = 1,000$ (APEX: $Re = 860,000$; $We = 305,000$)
- Similar results for horizontal upward-facing surface



Extra Slides



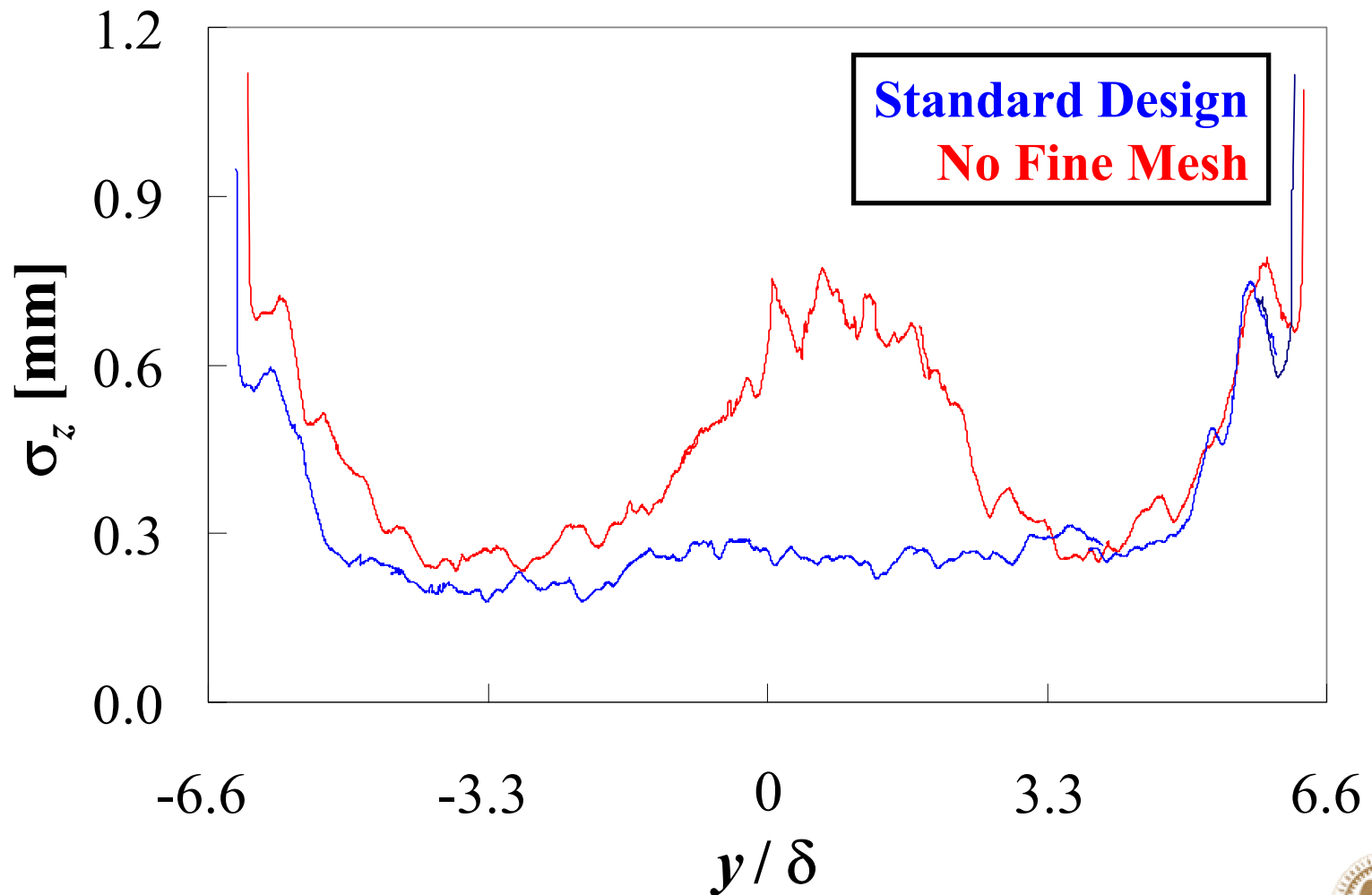
Effect of Cut Depth Δz_{cut} ($x / \delta = 16$)



- Surface ripple at jet center
- Cut jets appear “smoother”
 - σ_z independent of cut depth for $\Delta z_{\text{cut}} \geq 0.38$ mm
 - Average values:
 - $\sigma_z = 0.238$ mm for $\Delta z_{\text{cut}} = 0$
 - $\sigma_z = 0.137$ mm for $\Delta z_{\text{cut}} = 0.38$ mm
 - $\sigma_z = 0.135$ mm for $\Delta z_{\text{cut}} = 0.5$ mm
- Surface ripple reduced by ~40%



Flow Conditioning Effects ($x / \delta = 15$)



BL Cutting Effects

$(x / \delta = 15; \Delta z_{\text{cut}} = 0.25 \text{ mm})$

