Cryogenic Implosion Experiments on OMEGA



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Summary

High areal densities have been measured in cryogenic target implosions on OMEGA

 Ignition-relevant cryogenic target implosions are being carried out on OMEGA

- The OMEGA Cryogenic Target System has been improved to support the cryogenic experimental campaign
 - safe and reliable production: 8 D₂ and DT targets per month
 - improved target positioning: median displacement ~25 μ m
 - median ice roughness is 1.0 μ m for DT and 3.0 μ m for D₂
- The compressed fuel areal density correlates with the adiabat driven by the shape of the laser pulse
- Preheat is minimized to maintain the adiabat in the fuel

Measured near-1-D calculated areal density, 202 ± 7 mg/cm², highest ever hydrogenic areal density equivalent to 100 g/cm³ or 500× liquid density.



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The production of cryogenic targets is a safe and reliable operation that paces the experimental campaign

- Deuterium implosion experiments began in 2001
 - imploded 162 D₂ targets
 - three-day production cycle
 - provide up to eight cryogenic targets per week
- Tritium implosion experiments began in 2006
 - imploded 36 cryogenic DT targets (D:T, 45:55)
 - targets are filled by permeation (no fill tube); requires 6000 Ci T_2 .
 - safe operation: facility emissions <6 Ci/yr

Improvements in the ice-layer quality and target position proceed in parallel with implosion experiments.

Cryogenic targets meet most of the requirements

- Routinely provide DT-ice layers with a roughness <1.0- μ m rms
 - D₂-ice rms roughness <3.0 μ m>, best is 1.3 μ m, the source of the problem is understood
- Require 12 h to form a DT-ice layer
 - 60% probability of achieving the smoothness specification within 24 h
 - cooling the target 0.8 K (for D_2) and 0.25 K (for DT) roughens the ice
- Gas density and ice roughness are known at the time of the implosion
- Decay of tritium to form ³He affects the ice smoothness
 - no evidence of He bubbles is observed.

The ice layer roughness is characterized from an optical shadowgraph

 Optical system to locate the gas–ice interface (precision $\pm 0.1 \ \mu m$)



• Numerical analysis:



2-D

fits

100

Fourier

The layering process is very repeatable: melting and reforming the ice layer using the same protocol yields a similar roughness



- Variability in the ice roughness is attributed to the variability in the initial location of the seed crystal and how it grows.
- The average ice roughness of 25 targets is 1.1±0.4 $\mu m.$ The median value is 1.0 $\mu m.$

Ice roughness results from two sources

Thermal perturbation by the target support (D₂ targets only)

- affects the low modes
- addressed only by design changes

Cracks in the ice

- distort the shadowgraph
- less precise analysis
- eliminated by controlling the solidification process

Orthogonal views of a DT target with a circular defect







8- μ m-deep crack; \simeq to a 25- μ K temperature gradient on the surface

The roughness of a DT-ice layer increases with time; there is no similar behavior with D₂ targets

 Effect is attributed to the decay of tritium and accumulation of ³He; there is no evidence of He bubbles in the ice layer.



The ice roughness is confirmed at the implosion and the gas density is inferred by experiments and calculations

 Rapid shroud retraction limits target exposure to <0.1 s: layer quality is unaffected



• Observing how rapidly the target melts allows the heat load into the target to be calculated (~0.17 mW)

- Calculations replicate the observed behavior
 - extrapolation to the first 0.1 s exposure show no change in gas density





The DT ice layer in a foam capsule possessed a greater roughness than typically obtained in non-foam capsules



Transparent ice layers in foam targets are achieved by melting and refreezing the ice layer to reduce the void content.

Implosion experiments are demonstrating compression of cryogenic hydrogen fuel to high areal densities

Target designs are being refined based upon these experiments



- Conducted a systematic scan of fuel adiabat and drive intensities
 - 2 < α < 10; α = fuel pressure/Fermi-degenerate pressure
 - $I_L = 2.5 \times 10^{14} \,\text{W/cm}^2$ to $1.5 \times 10^{15} \,\text{W/cm}^2$
 - $V_{\rm imp} = 2.5 \times 10^7$ to 4.0×10^7 cm/s
 - In flight aspect ratio: 30–50
 - Number of perturbation e-folds ~2 to 5
- Areal density was inferred from the energy loss of energetic protons*

Measured areal densities agree with calculated values when shock velocity is adequately modeled and hot-electron preheat mitigated.

Areal density depends mainly on the in-flight fuel adiabat



- accuracy for the laser absorptio during picket $\Delta E/E \sim 10\%$

Phys. Rev. Lett. <u>34</u>, 721 (1975).

^{*} R. Betti and S. Zhou, Phys. Plasma <u>12</u>, 110702 (2005).

^{**}R. C. Malone, R. L. McCrory, and R. L. Morse,

Preheating by energetic electrons from the two plasmon decay increases rapidly with laser intensity

- Shocks adequately timed to not compromise adiabat
- Hot-electron preheat inferred using measured hard-x-ray signals produced by Bremsstrahlung radiation from fast electrons



Fuel preheat from energetic electrons is reduced by increasing the CD thickness due to the lower two plasmon decay instability threshold in D₂.

Effect of shock timing on the adiabat is tested by changing the pulse shape



Modeling resonance absorption and nonlocal transport is essential for comparing with experimental results

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Using an optimal pulse shape and ablator thickness allowed the fuel assembly to proceed according to the 1-D calculation



Future experiments, and those in progress, will address conditions required for ignition

- Increase the implosion velocity to 3.5×10^7 cm/s
 - thinner ice layer (65- μ m D₂)
- Alternative methods to minimize energetic-electron preheat and reduce RT growth

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